

IN THE CLAIMS:

Please amend the claims, as follows:

Claim 1. (Currently Amended): A combustion method comprising steps of:
forming combustion air so as to have a jet flow cross section having a larger specific surface area than that in a case of supplying from a circular throat the same quantity of combustion air as said combustion air has; injecting said combustion air into a furnace having a furnace temperature of not less than 800°C; injecting fuel toward a jet flow of said air in the furnace; entirely colliding a jet flow of said fuel with said air jet flow within a range of positions where [[that]] said fuel jet flow has sufficient turbulent strength and at a position where oxygen density of said air jet flow becomes low; and rapidly mixing said fuel with said air jet flow with [[strong]] turbulences at collision.

Claim 2. (Original): A combustion method according to claim 1, wherein said combustion air is supplied after being preheated to have a high temperature close to a temperature of combustion exhaust gas by collecting heat of said combustion exhaust gas exhausted from a regenerative medium.

Claim 3. (Previously Presented): A combustion method according to claim 1, wherein said combustion air is formed into a jet flow which is flat and has a thin radial thickness as a whole.

Claim 4. (Previously Presented): A combustion method according to claim 1, wherein said fuel is injected from at least two separate injection openings, caused to collide with an air jet flow having an increased specific surface area in a wide area, and

rapidly mixed with said air jet flow with strong turbulences.

Claim 5. (Previously presented): A combustion method according to claim 1, wherein a plurality of said fuel jet flows are formed, and said fuel jet flows collide with each other in the furnace before coming in contact with said air jet flow.

Claim 6. (Previously presented): A combustion method according to claim 1, wherein a plurality of said air jet flows are formed, and said air jet flows collide with each other in the furnace before coming in contact with said fuel jet flow.

Claim 7. (Previously presented): A combustion method according to claim 1, wherein a plurality of said fuel jet flows and said air jet flows are formed, and said air jet flows collide with each other and said fuel jet flows collide with each other in the furnace before said fuel jet flows collide with said air jet flows.

Claim 8. (Previously Presented): A combustion method according to claim 1, wherein a plurality of pairs of said fuel jet flows and said air jet flows which collide with each other in said furnace are formed in order to form a large combustion field.

Claim 9. (Previously Presented): A combustion method according to claim 1, wherein said combustion air whose quantity is less than a theoretical air quantity is supplied and high-temperature air combustion for forming a non-oxidizing atmosphere or a reduction atmosphere is carried out.

Claim 10. (Previously Presented): A combustion method according to claim 1, wherein said combustion air whose quantity is not less than the theoretical air quantity is supplied and high-temperature air combustion by turbulent diffusion mixing is provoked.

Claim 11. (Currently Amended): A burner comprising: an air throat for forming a combustion air so as to have a jet flow cross section having a larger specific surface area than that in a case of supplying from a circular throat the same quantity of combustion air as said combustion air has and for injecting a full quantity of said combustion air into a furnace having a furnace temperature of not less than 800°C; and a fuel nozzle for injecting fuel into said furnace so as to cause a jet flow of said fuel to entirely collide with said air jet flow at a position away from an injection opening of said air throat by a specific distance before losing velocity energy of said fuel jet flow.

Claim 12. (Original): A burner according to claim 11, wherein said air throat has a flat rectangular opening.

Claim 13. (Original): A burner according to claim 11, wherein said specific surface area can be increased by dividing said air throat into a plurality of small holes.

Claim 14. (Original): A burner according to claim 11, wherein said air throat is divided into a plurality of small holes, and respective jet flows are continuously arranged in a form of a line without being independent to form a jet flow having a flat cross-sectional shape as a whole.

Claim 15. (Previously Presented): A burner according to claim 13, wherein a plurality of said small holes form a jet flow in which said air jet flows collide with each other before coming in contact with said fuel jet flow.

Claim 16. (Previously Presented): A burner according to claim 11, wherein said fuel nozzle has at least two injection openings and causes said fuel to collide with said air jet flow having an increased specific surface area in a wide area.

Claim 17. (Previously Presented): A burner according to claim 11, wherein said fuel nozzle has at least two injection openings and forms a jet flow in which fuel jet flows injected from said respective injection openings collide with each other before coming in contact with said air jet flow.

Claim 18. (Previously Presented): A burner according to claim 11, wherein a plurality of said fuel nozzles are arranged so as to surround said air jet flow.

Claim 19. (Original): A burner according to claim 18, wherein a plurality of said fuel nozzles form a jet flow in which said fuel jet flows collide with each other before coming in contact with said air jet flow.

Claim 20. (Previously Presented): A burner according to claim 11, wherein a plurality of said air jet flows and a plurality of said fuel jet flows are formed, and said air jet flows collide with each other and said fuel jet flows collide with each other before said air jet flows collide with said fuel jet flows.

Claim 21. (Previously Presented): A burner according to claim 11, wherein said air throat includes a regenerative medium and flow switching means for alternately leading combustion exhaust gas and combustion air to said regenerative medium, and injects into said furnace said combustion air preheated to have a high temperature close to a combustion exhaust gas temperature through said regenerative medium.

Claim 22. (Original): A burner according to claim 21, wherein a ceramic honeycomb is included as said regenerative medium.

Claim 23. (Previously Presented): A burner according to claim 21, wherein said regenerative medium is included in said air throat, and said flow switching means is directly connected to a burner body constituting said air throat and switches said combustion air and said exhaust gas at a short distance from said regenerative medium.

Claim 24. (Previously Presented): A burner according to claim 11, wherein said combustion air supplied from said air throat has a quantity which is less than a theoretical air quantity, and high-temperature air combustion for forming a non-oxidizing atmosphere or a reduction atmosphere is carried out.

Claim 25. (Previously Presented): A burner according to claim 11, wherein said combustion air supplied from said air throat has a quantity which is not less than the theoretical air quantity and high-temperature air combustion by turbulent diffusion mixing is provoked.

Claim 26. (Original): A burner according to claim 24, wherein said fuel is injected with a ratio d_e/D_{pcd} of a corresponding diameter d_e of an opening of said air throat and a gap $1/2D_{pcd}$ from a center of said air throat to a center of said fuel nozzle falling within a range of 0.1 to 0.5 and with a ratio L_a/d_e of said corresponding diameter d_e of said air throat relative to a distance L_a from an intersection of a fuel injection axis and a plane on a central axis in a longitudinal direction of said air throat to an outlet surface of said air throat falling within a range of 1.0 to 5.0.

Claim 27. (Original): A burner according to claim 25, wherein said fuel is injected with a ratio d_e/D_{pcd} of a corresponding diameter d_e of an opening of said air throat and a gap $1/2D_{pcd}$ from a center of said air throat to a center of said fuel nozzle falling within a range of 0.1 to 0.5 and with a ratio L_a/d_e of said corresponding diameter d_e of said air throat relative to a distance L_a from an intersection of a fuel jet flow axis and a plane on a central axis in a longitudinal direction of said air throat to an outlet surface of said air throat falling within a range of 2.0 to 10.0.

Claim 28. (Previously Presented): A burner according to claim 11, a plurality of pairs of said air throats and said fuel nozzles are set to constitute a large combustion machine.